

**IN THE CLAIMS:**

Claims 17, 21, 24, 27, 29 and 31 have been amended. Claims 20 and 30 have been canceled. New claims 32 and 33 have been added. The following listing of claims replaces all prior versions and listings of claims in the application.

**Listing of Claims:**

Claims 1 to 16 (cancelled).

Claim 17 (currently amended): A method for manufacturing vane segments for a gas turbine comprising the steps of:

providing a plurality of vanes,  
manufacturing a vane segment from the plurality of vanes via powder metallurgy injection molding, the step of manufacturing including the steps of:  
mixing a metal powder having a binding agent to form a homogeneous material;  
forming at least one molded body from the homogeneous material via injection molding,  
subjecting the at least one molded body to a debinding process, and  
compressing the at least one molded body via sintering to form the vane segment.

Claim 18 (previously presented): The method as recited in Claim 17, wherein the vane segment is designed as a guide vane segment and includes at least two guide vanes.

Claim 19 (previously presented): The method as recited in Claim 18, wherein the guide vane segment includes three or four guide vanes.

Claim 20 (canceled).

Claim 21 (currently amended): The method as recited in Claim ~~20~~ 17, wherein, in that in order to manufacture one vane segment from at least two vanes,  
a molded body for each vane is manufactured separately via injection molding, and  
the molded bodies are joined together prior to the debinding process to form one molded body for the vane segment.

Claim 22 (previously presented): The method as recited in Claim 21, wherein the molded bodies for the vanes are joined together prior to the debinding process in the green state to form one molded body for the vane segment.

Claim 23 (previously presented): The method as recited in Claim 21, wherein the one molded body for the vane segment is then subjected to a uniform debinding process and uniform sintering in the debinding and sintering steps.

Claim 24 (currently amended): The method as recited in Claim ~~23~~ 17, wherein, in that to manufacture one vane segment from at least two vanes:

a molded body for each vane is manufactured separately via injection molding,  
the molded bodies for the vanes undergo separate debinding processes, and  
the molded bodies for the vanes are subsequently joined together to form one molded body for the vane segment.

Claim 25 (previously presented): The method as recited in Claim 24, wherein the molded bodies for the vanes are joined together in a presintered state to form one molded body for the vane segment.

Claim 26 (previously presented): The method as recited in Claim 24, wherein the one molded body for the vane segment is then subjected to uniform sintering in the sintering step.

Claim 27 (currently amended): The method as recited in Claim ~~20~~ 17, wherein, to manufacture one vane segment from at least two vanes, a joint molded body for all vanes of the vane segment is manufactured via injection molding.

Claim 28 (previously presented): The method as recited in Claim 27, wherein the joint molded body for the vane segment is subjected to a uniform debinding process and uniform sintering.

Claim 29 (currently amended): A component for a gas turbine, comprising  
a guide vane segment manufactured from a plurality of guide vanes via powder

metallurgy injection molding.

Claim 30 (canceled).

Claim 31 (currently amended): The component as recited in Claim ~~30~~ 29, wherein the guide vane segment includes three or four guide vanes.

Claim 32 (new): The method as recited in claim 29 wherein the plurality of guide vanes are connected via an inner cover band and an outer cover band.

Claim 33 (new): A method for manufacturing guide vane segments for a gas turbine comprising the steps of:

providing a plurality of guide vanes,

manufacturing a guide vane segment from the plurality of vanes via powder metallurgy injection molding.